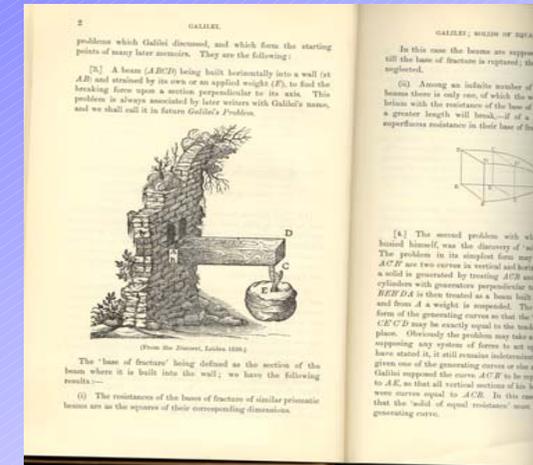


Round table on Multicomponent measurement to improve the traceability in the force chain



Carlo Ferrero

INRiM Consultant

IMEKO TC8 Chairman

*EA – Chairman of the
“Mechanical Commission”*



During the last meeting of the TC3 (IMEKO World Congress, Lisbona, September 2009), was decided to organize this Round Table on Multicomponent measurement to improve the traceability in the force chain

I am very glad to have this opportunity together my colleagues and friend that are the other speakers

Design and analysis of a column type multi-component force / moment sensor

*** J.H. Kim, D.I. Kang , H.H. Shin, Y.K. Park**

***Division of Physical Metrology/Mass and Force Group , KRISS ,P.O . Box
102, Yusong-Gu , Daejeon , 305- 600, South Korea***

Abstract

A sensing element of column type was devised as a multi-component force /moment sensor by attaching strain gages. The ratio of length over diameter (L/D) for the sensing element was designed analytically and verified by finite element analysis.

In order to reduce the interference error of each loading component, this paper proposed a decoupling method with the addition and subtraction processes using signals of strain gages. Finally the calibration showed that the interference error of F component was less than 7.3% FS, and in case of other components, 5.0% FS.

J.W. Joo, S.Y. Jeoung, G.S. Kim, D.I. Kang, Evaluation of interference error for six-component load cell, in: KSME Annual Fall Conference 96F074 (1996) 441-446. 2.

D.I. Kang, G.S. Kim, S.Y. Jeoung, J.W. Joo, Design and evaluation of binocular type six-component load cell by within 7.6%. The interference error of F com- x using experimental technique, KSME 21 (11) (1997) 1921–

G.S. Kim, D.I. Kang, S.Y. Jeoung, J.W. Joo, Design of sensing element for 3-component load cell using parallel plate structure, KSME 21 (11) (1997) 1871–1884.

D.I. Kang, Design and application of force measuring system using build-up technique, Ph.D. Dissertation, Dept. of Mech.Engng, KAIST, 1994.

**EVALUATION OF MULTI-COMPONENT FORCE TRANSDUCERS HAVING
 COLUMN TYPE SENSING ELEMENT**

*Yon-Kyu Park*¹, *Rolf Kumme*², *Dirk Roeske*³, *Dae-Im Kang*⁴



Fig. 1. Column-type multi-component force transducers.

They were calibrated statically with a force–moment calibration machine in the Physikalisch-Technische Bundesanstalt (PTB), Germany.

We evaluated the dynamic characteristics of the force–moment transducers by using a shaker system and a multi-channel dynamic analyser system.

Table 1. Capacity of the column-type force transducers.

Component	Solid transducer	Hollow transducer
F_x	1.5 kN	2 kN
F_y	1.5 kN	2 kN
F_z	20 kN	20 kN
M_x	40 N·m	60 N·m
M_y	40 N·m	60 N·m
M_z	60 N·m	90 N·m



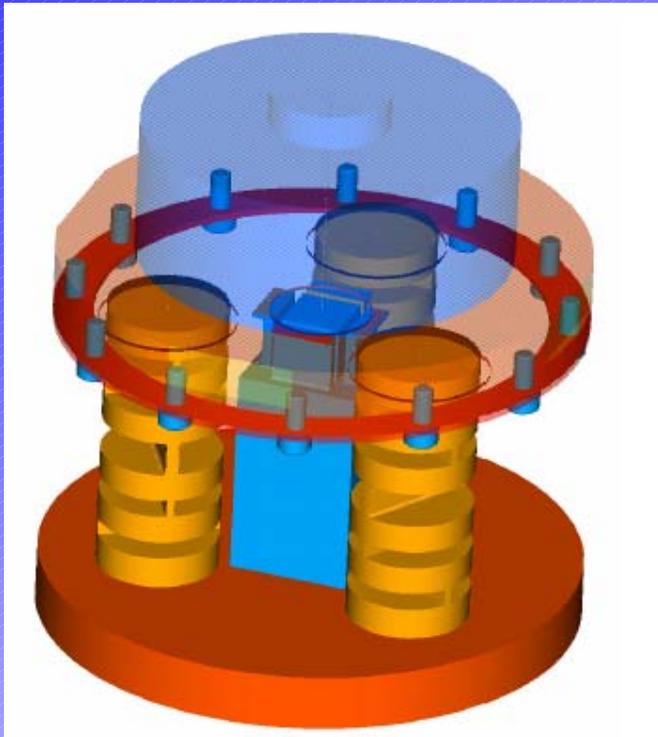
D. Roeske, “Metrological characterization of a hexapod for a multi-component calibration device,” *XVII IMEKO World Congress*, pp.347-351, Dubrovnik, Croatia, 2003.

Y.K. Park, R. Kumme and D.I. Kang, “Dynamic evaluation of column type multi-component force transducers,” *XVIII IMEKO World Congress*, Rio de Janeiro, Brazil, 2006.

Y.K. Park, R. Kumme and D.I. Kang, “Dynamic investigation of a binocular 6-component force–moment sensor,” *Meas. Sci. Technol.*, vol. 13, pp. 1311-1318, 2002.

INVESTIGATION AND CALIBRATION OF A FORCE VECTOR SENSOR WITH A CALIBRATION ARTEFACT

Sara Lietz 1, Falk Tegtmeier 1, Dirk Röske 1, Rolf Kumme 1, Daniel Schwind



Force vector sensor
mounted on the
spherical
calibration
artefact

S. Lietz, F. Tegtmeier, R. Kumme, D. Röske, U. Kolwinski, K. Zöllner, *A new six-component force vector sensor – first investigations*, Proc. 20th IMEKO TC3 Conference, Merida, Mexico, Nov 26-30, 2007.

S. Lietz, F. Tegtmeier, R. Kumme, D. Röske, D. Schwind, U. Kolwinski, *Darstellung und Messung von Kräften mittels vektorieller Kraftsensoren*, Proc. Sensoren und Messsysteme 2008, VDI-Berichte 2011, Ludwigsburg, March 11-12, 2008.

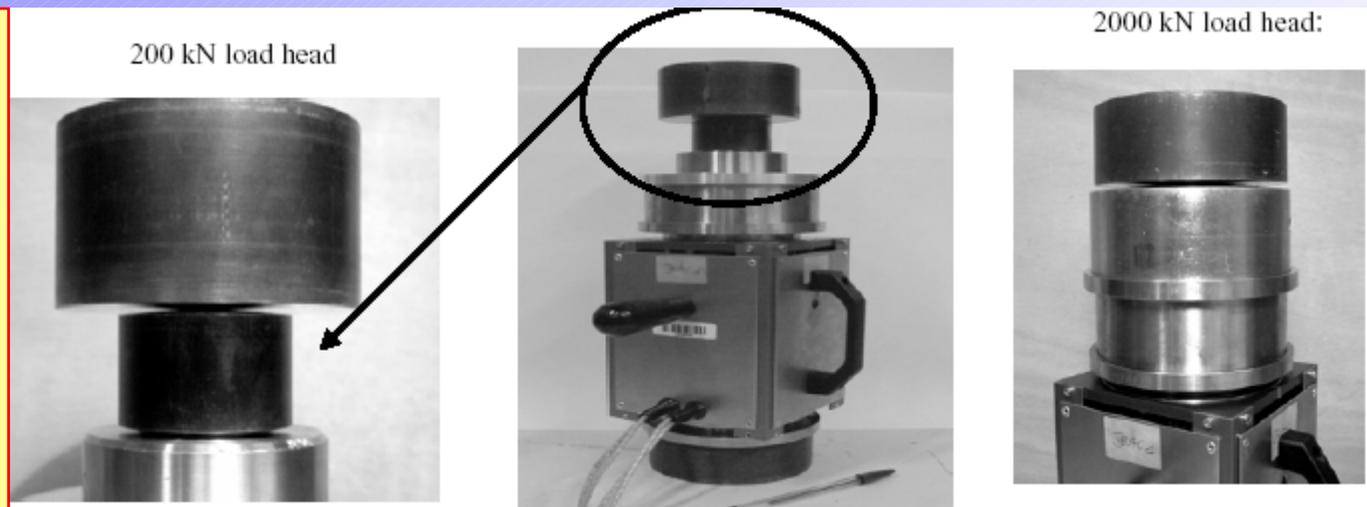
STRAIN CYLINDERS FOR COMPRESSION TESTING MACHINES

Philippe Averlant

ABSTRACT

Measurement results of the compressive strength of concrete specimens depend on the manner of force transfer by the testing machine. This performance is verified according to a standard using strain cylinders. However, the qualification or calibration of strain cylinders is not defined in the standard. This paper presents a new procedure for cylinder calibration. The measurand is defined.

The standard quantity is modelised depending on the specific manner of loading during calibration. Uncertainties are computed using theoretical equations. The results of a first cylinder calibration are given.



Round Table

Coordinator:

Dr. Carlo Ferrero

Istituto Nazionale di Ricerca Metrologica (INRIM)

E-mail: C.Ferrero@inrim.it

Speakers:

Carlo Ferrero:

Development and Calibration of the INRiM composite 6-components dynamometers and the results of 20 years of intercomparisons

Dae-Im Kang:

Evaluation of FSM by using build-up system

Rolf Kumme or Dirk Roeske:

Calibration of a force vector sensor with a multicomponent calibration system

Philippe Averlant:

Calibration of strain cylinders

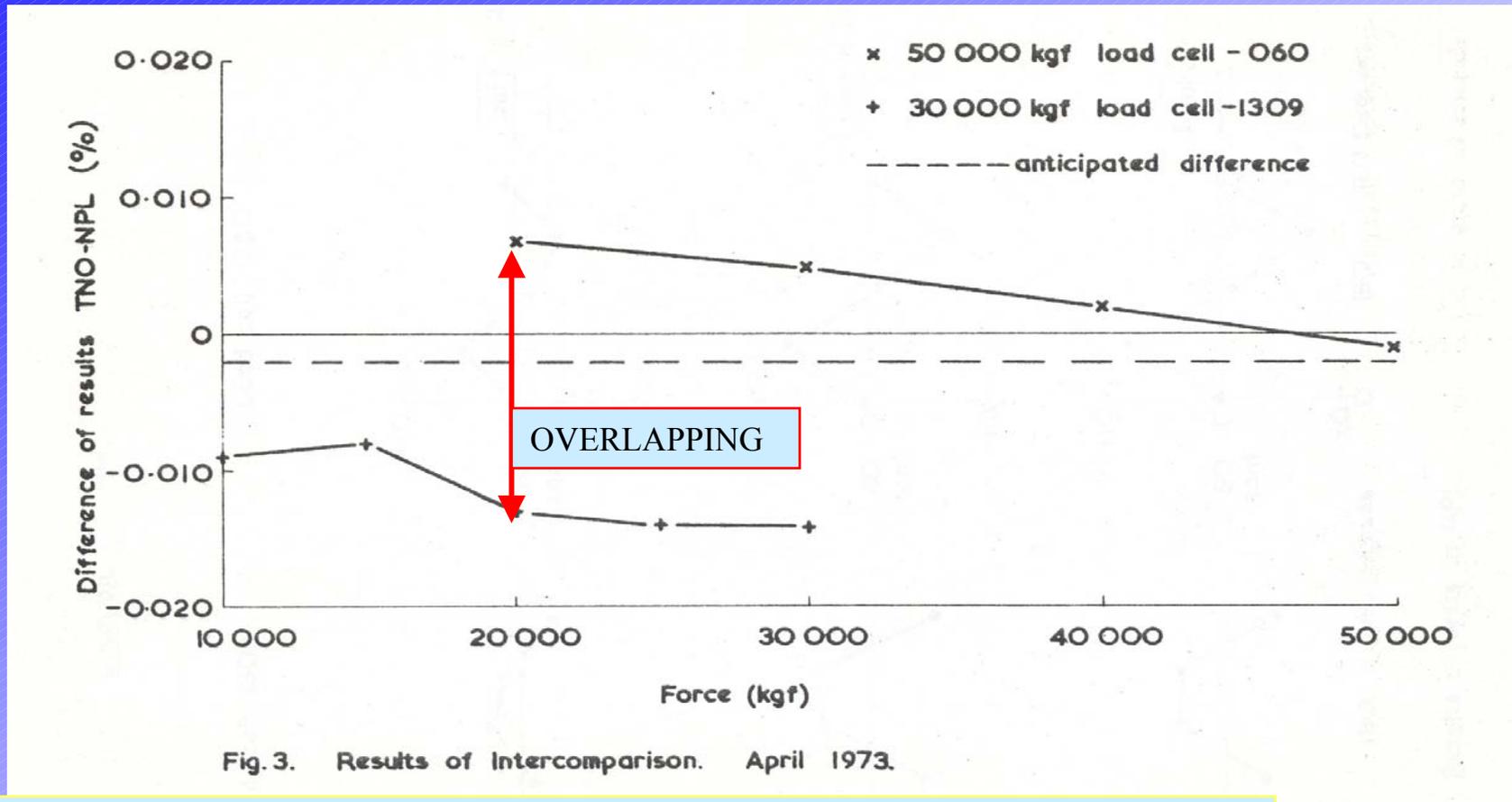
**Why 35 years of research in the field of
force multicomponents from the
beginning of the '70s ?**

From the Report of Prof. Weiler (PTB) after the 2° IMEKO TC3 (1971)

The 2nd discussion meeting of the Technical Committee Force and Mass" was organized by Dr. H. **Wieringa** in **The Hague, Netherlands, in September 1971.**

During the discussion of lectures **Prof. A. Bray (IMGC, Torino, Italy)** proposed to found an ad hoc committee in order to organize comparison measurements of force standard machines of the different government institutes. The ad hoc committee agreed that **Prof. W. Weiler (PTB, Braunschweig, GFR)** should start this work by comparing the small dead weight machines for 20 kN and 100 kN where as **Mr. Debnam (NPL)** and **Mr. Wieringa (TNO)** should compare the 0.5 MN machines of the Netherlands and of the United Kingdom.

Debnam, R.C., Wieringa H., An intercomparison of force standard machines, VDI-Berichte No 212, 1974, pp. 122-132.



Results of the intercomparison made in April 1973

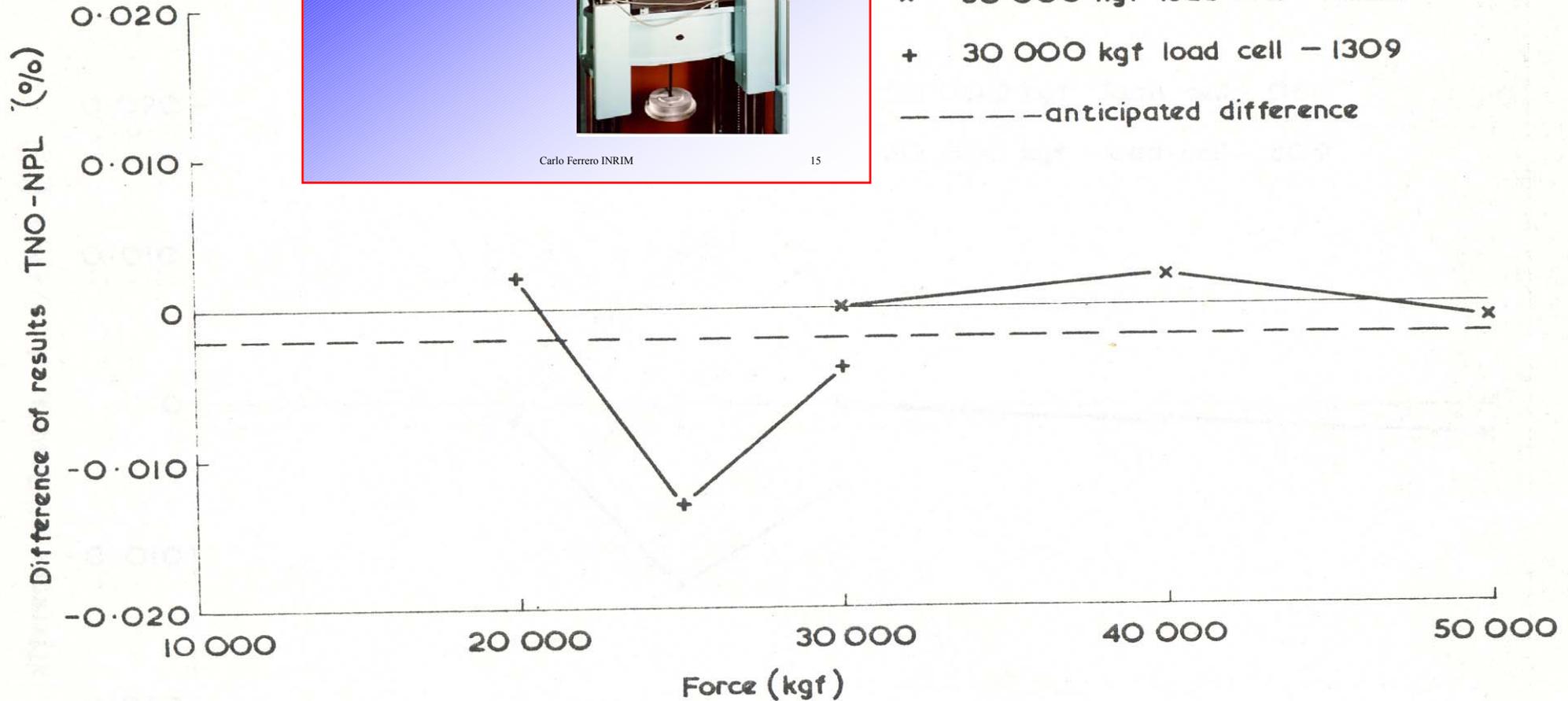
NPL-Machine



Carlo Ferrero INRIM

15

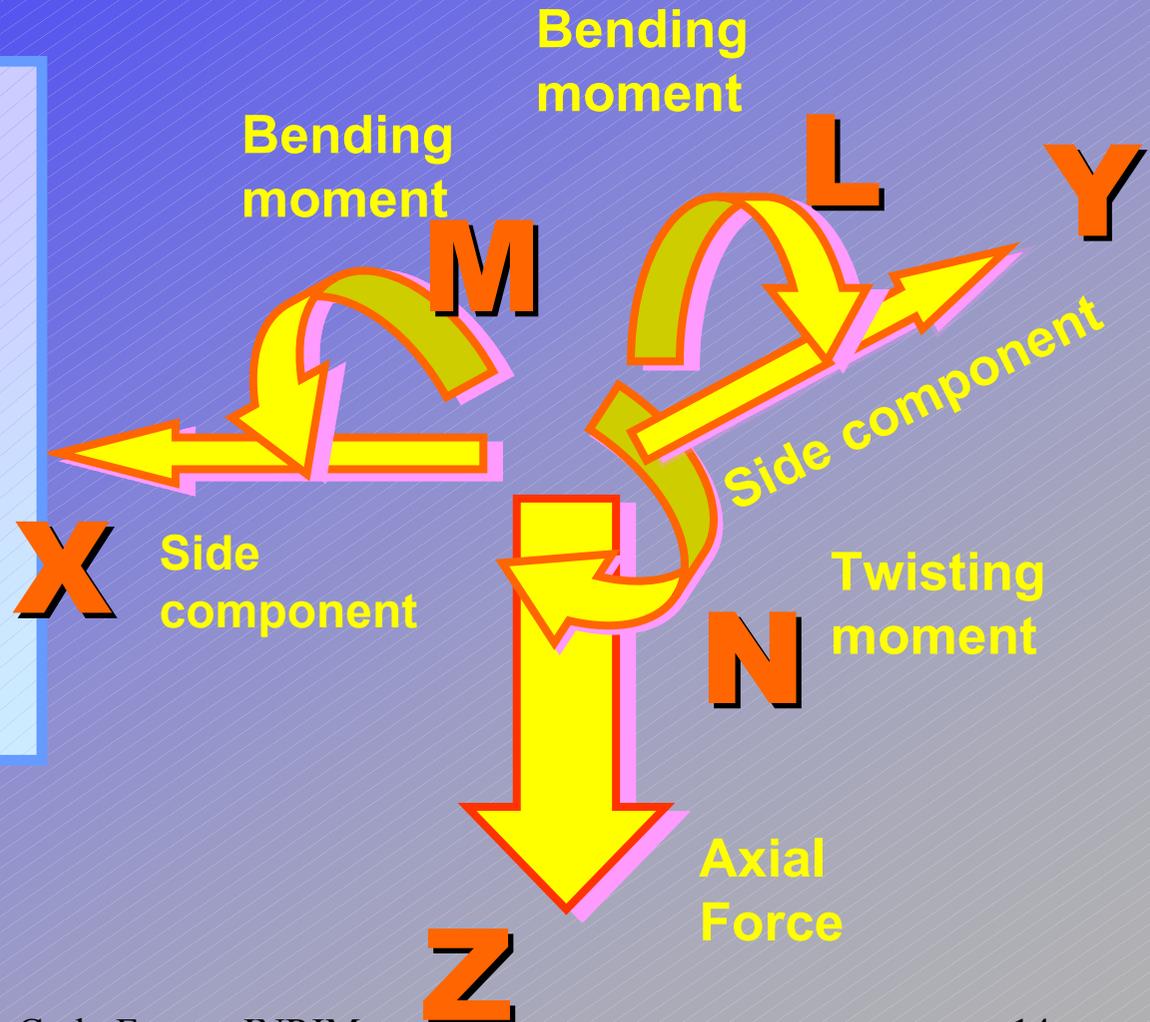
- x 50 000 kgf load cell - 060
- + 30 000 kgf load cell - 1309
- anticipated difference



Result of the intercomparison repeated in May 1975 after to have to check the vertical position of the DWM

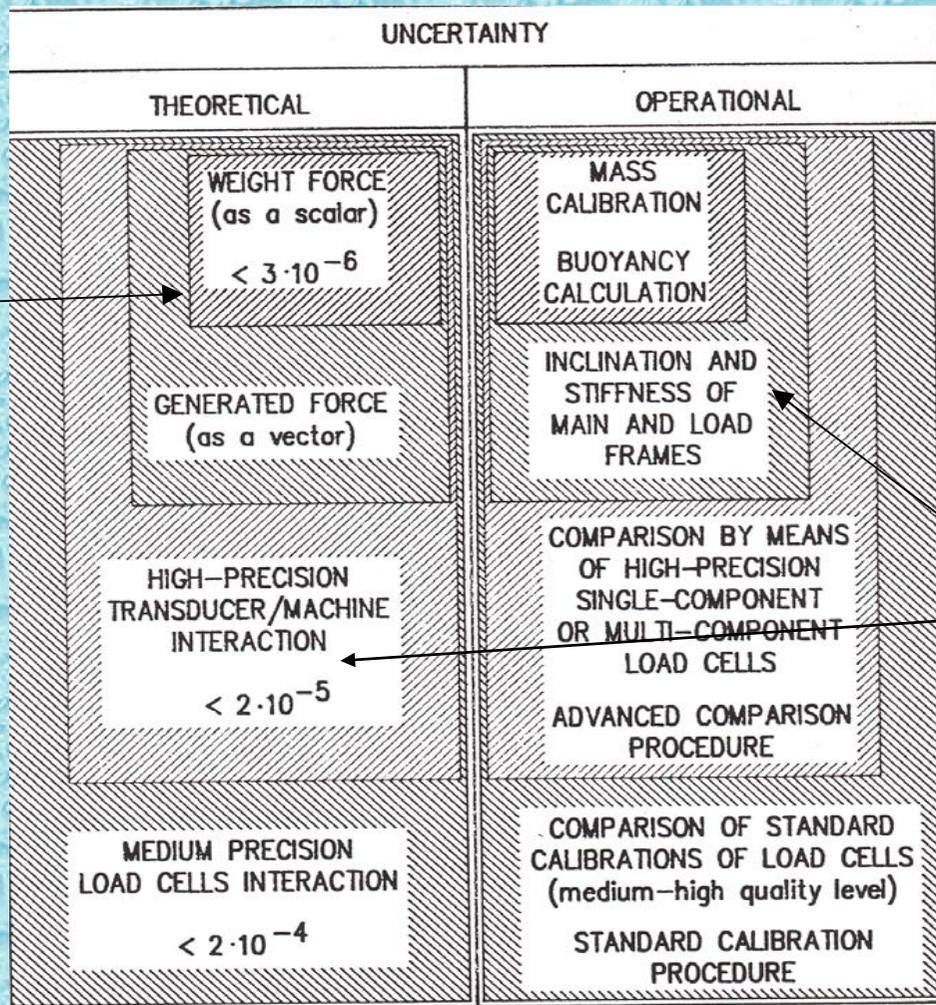
Ideally, force standard machines should apply uniaxial loads only.

In practice, mis-alignments and deformations of the force standard machines result in finite values of the **five parasitic components** of the force/moment tensor.



Several international comparisons of primary force standards demonstrated that the uncertainty of the axial-force value standards is sometimes higher by one order of magnitude than the uncertainty that might be expected on the basis of the relative uncertainties of the values of mass, of the acceleration due to gravity and of Archimede's thrust.

SCALAR



Vectorial

FORCE IS A VECTORIAL QUANTITY

A number of **International comparisons** were carried out with single-component load cells

starting from 1973 up to now.

These comparisons, which evidenced a **rotational effect and overlapping phenomena** caused by the **interaction of the cells with the machines**, made great improvements possible as regards reduction of the uncertainty in axial load determination (from a few 10^{-4} to some parts in 10^{-5})**and**.....

.....**and** showed that a multi component dynamometer (Dubois et al, 1980: Ferrero et al, 1981) is an essential tool in order

- to **improve force standard** deadweight machines,
- to attempt to **explain anomalies**,
- to **optimize testing methods**, and
- to **give manufacturers indications** allowing them to **improve both force standard machines and load cells**

**In order to improve primary force standards as well as to understand anomalies and optimise calibration methods,
it is essential to measure the effect of different parameters on the parasitic components**

**Than to design and to realize
six-component dynamometers
to measure
parasitic components generate
by Force standard machines**



Commission of the European Communities

bcr information

Applied Metrology

THE MEASUREMENTS OF PARASITIC COMPONENTS
IN THE IMG C, NPL, TNO, LNE
AND PTB FORCE STANDARD MACHINES
BY MEANS OF THE IMG C
SIX COMPONENT 100 KN DYNAMOMETER

**Second BCR-IMG C contract for
the first European intercomparisons**



Report
EUR 12457 EN

Blow-up from microfiche original

Five deadweight force standard machines (DWM) belonging to countries of the European Communities (NPL, TNO, PTB, LNE, IMG C), were checked by means of the IMG C dynamometer from 1983 to 1985.

**Dubois, M., Bourateu, J.P., Gosset, A., Priel, M.,
Intercomparison de trois bancs d'étalonnage dynamométrique
de capacité 250-300 kN, Bulletin du BNM, 11, 1980, pp. 17-27**

Preliminary Results

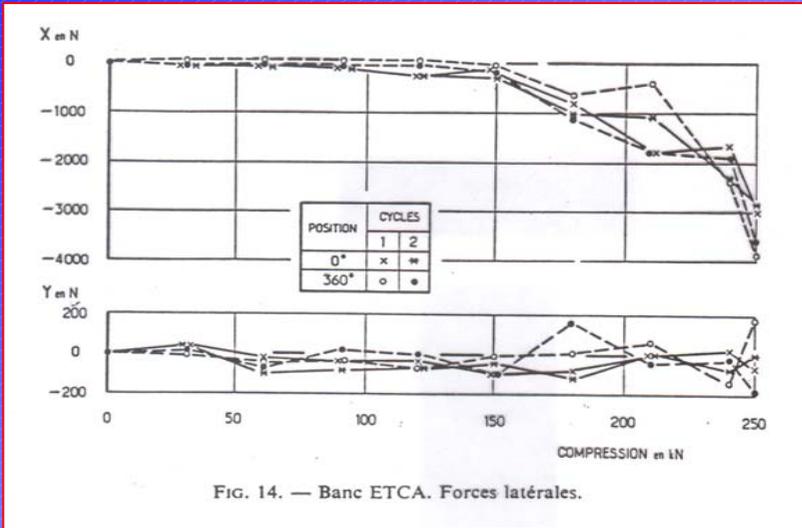


FIG. 14. — Banc ETCA. Forces latérales.



**ONERA 4-
component
single block
dynamometer**

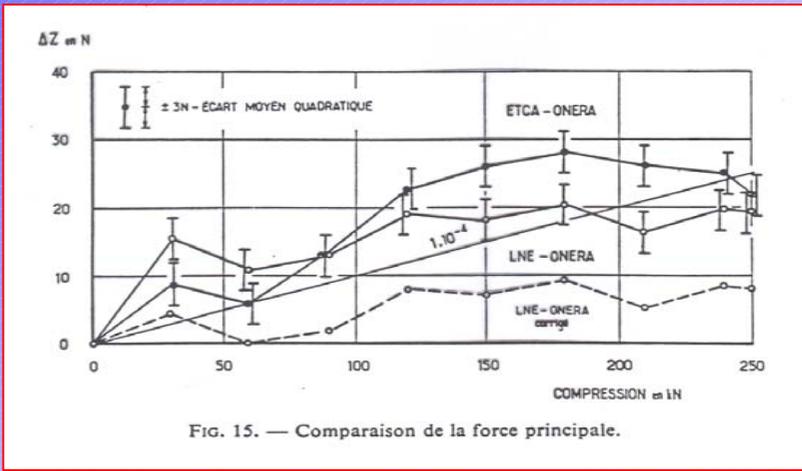


FIG. 15. — Comparaison de la force principale.

Another contract was signed by LNE by using the ONERA 4-components dynamometer.

The standards machines of NPL, TNO, LNE were checked also by means of the ONERA 4-component dynamometer, with the aim of:

1. bringing into evidence a **possible influence of the different stiffness of the two dynamometers** on the metrological characteristics of these standard machines, as well as
2. determining the **repeatability and the reproducibility both of primary force standards and transfer standard dynamometers.**



Commission of the European Communities

BCR Information

Applied Metrology

EVALUATION OF
FORCE STANDARD MACHINES
WITH TWO
MULTI-COMPONENT DYNAMOMETERS



Report
EUR 12726 EN

Blow-up from microfiche original

**3rd contract BCR-
IMGC for the
evaluation of the results
obtained with the 2
different dynamometers
(integral type or
composite type) on
different DWM**

Gosset A, Nossent P 1986 *Mesure de plusieurs machines de force étalons à l'aide d'un dynamomètre multicomposantes de capacité 300 kN en compression.* BCR Technical Report. Laboratoire National d'Essais, Paris

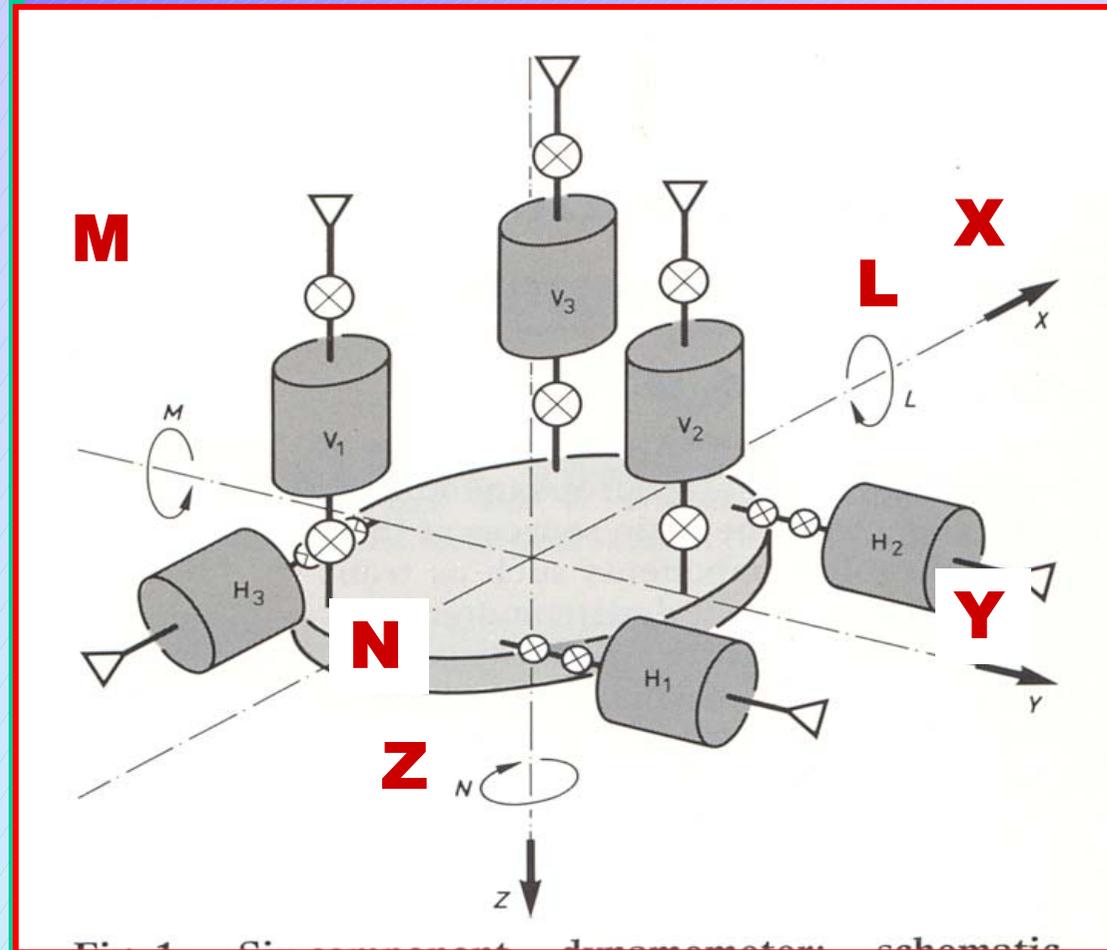
THE SIX-COMPONENT DYNAMOMETER, THE MEASUREMENT SET-UP AND PROCEDURES

The INRiM six-component dynamometers were designed and constructed with the purpose of measuring, in addition to axial load, also the five parasitic components (i.e., transverse forces and moments).

INRIM dynamometer **is a composite load cells consisting of six uniaxial load cells arranged to measure the:**

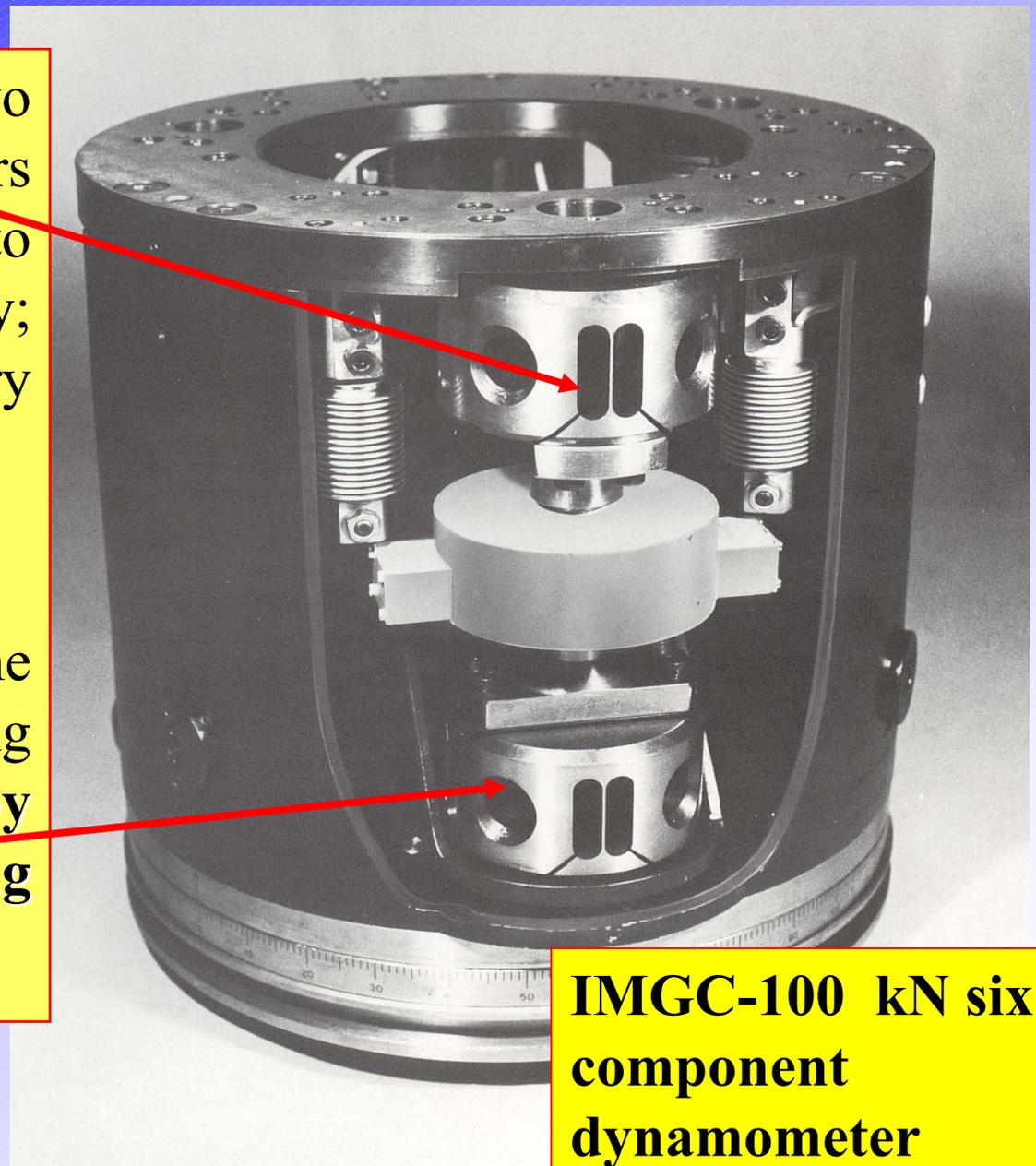
- **Vertical load Z;**
- **Side forces X and Y;**
- **Bending moments L and M;**
- **twisting moments N.**

Decoupling between the load cells is provided **by the use of elastic flexures**



The **elastic flexure** at the two ends of composite dynamometers allow **mechanic decoupling** to be made very effectively; interactions are therefore very weak.

Here the elements measuring the axial component and bending moments **work in tension** by means of ~~double~~ **decoupling end-flexures**.



IMGCC-100 kN six component dynamometer

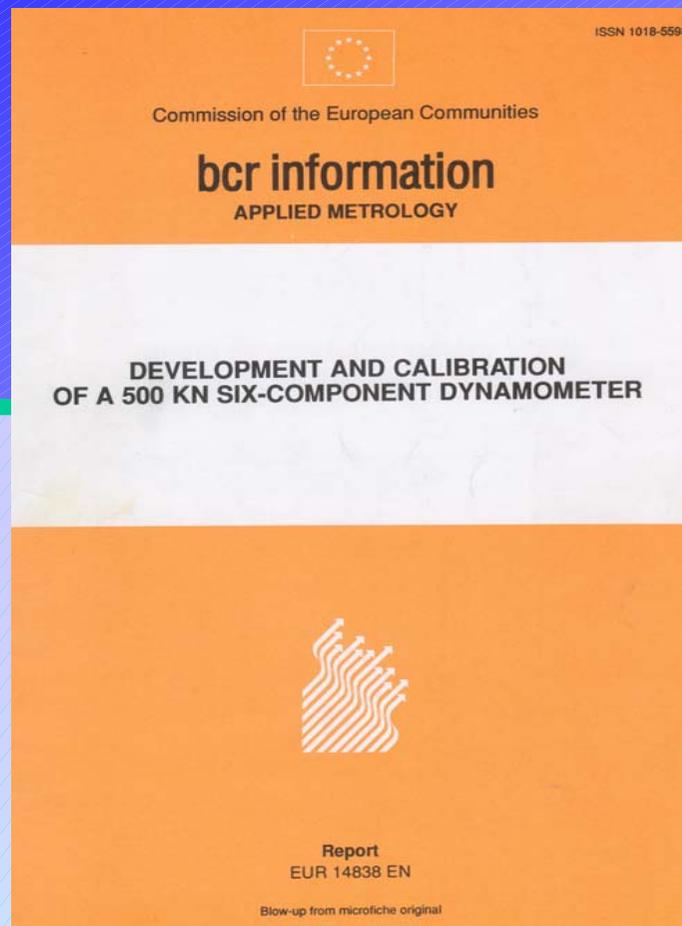
The **composite configuration** has several advantages with respect to other types of multi-component dynamometers:

a) **high sensitivity** to transverse components and to twisting moment

b) **low interaction** between the axial component and the transverse components

c) **low dependence** on interface conditions and, consequently, on parasitic components during the load transmission phase

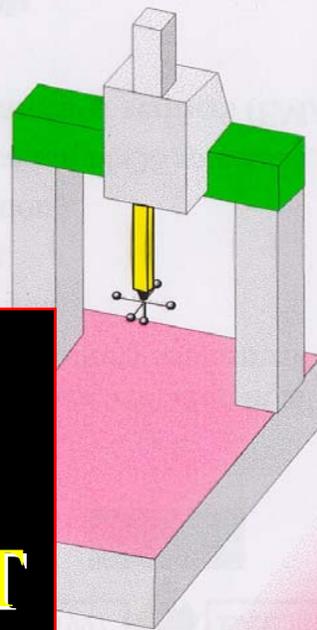




Quarto contratto BCR-IMGC per la realizzazione del nuovo dinamometro

Following the success of this work, the European Union commissioned IMGC to design and construct a **500 kN six-component dynamometer**





*a measuring-instrument
with unknown
measuring uncertainty
is only a piece of
furniture*

HOW TO CALIBRATE THE MULTICOMPONENT DYNAMOMETER?





Commission of the European Communities

BCR information

APPLIED METROLOGY

AR. 374 (vol. R. Tecnica R 181)

ANALYSIS AND CALIBRATION OF IMG6 SIX-COMPONENT DYNAMOMETER

(Technical Report)

Volume I



Report
EUR 10229 EN/1



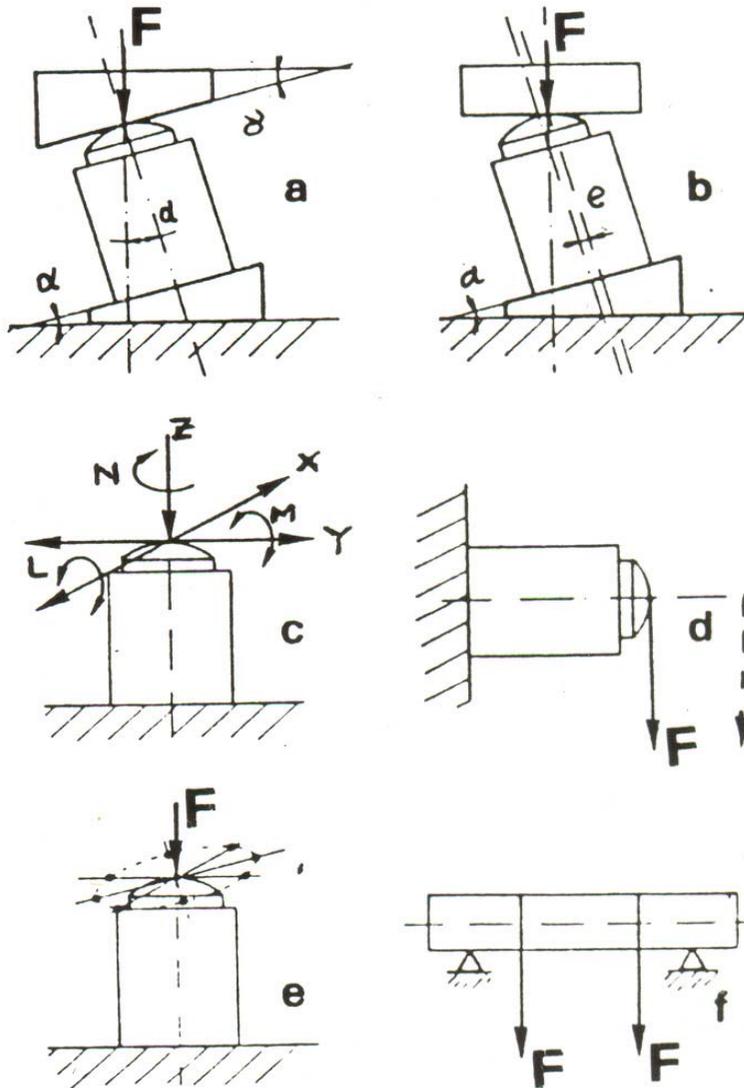
First BCR-IMG6 contract for the calibration of the IMG6 100 kN 6-components dynamometer with the ONERA calibration system

SISTEMI DI TARATURA INRIM



CHARACTERIZATION OF LOAD CELLS BY MEANS OF DIFFERENT CALIBRATION SYSTEMS AT IMGC

Different TESTING METHODS AND APPARATUS evaluated by IMGC



The calibration methods can schematically be outlined as follows:

a) **Two-wedge method:** side components (X, Y) are proportional to axial load Z and functions of inclination angle β ;

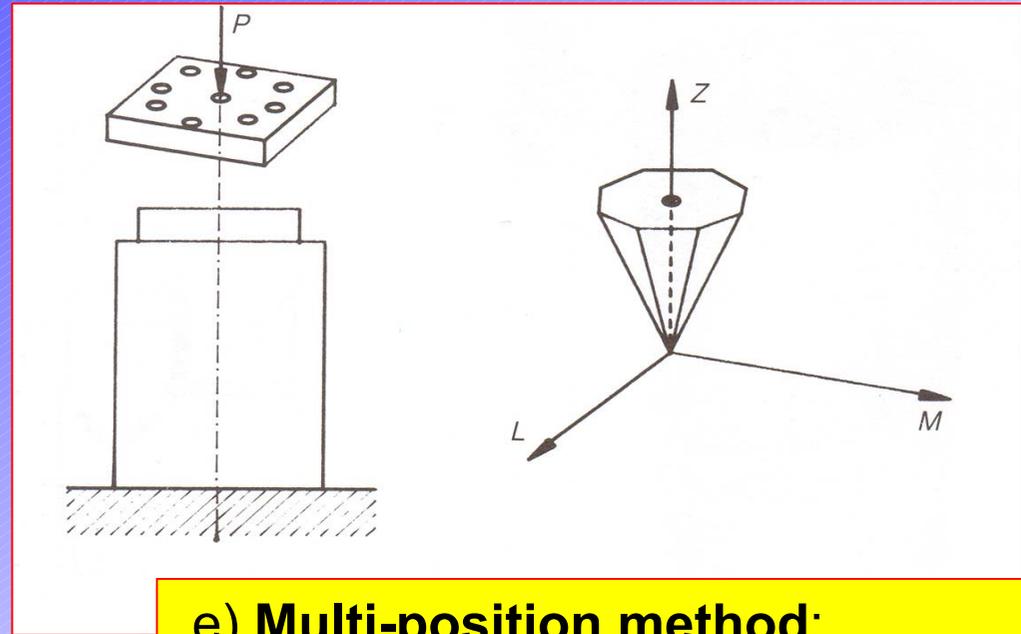
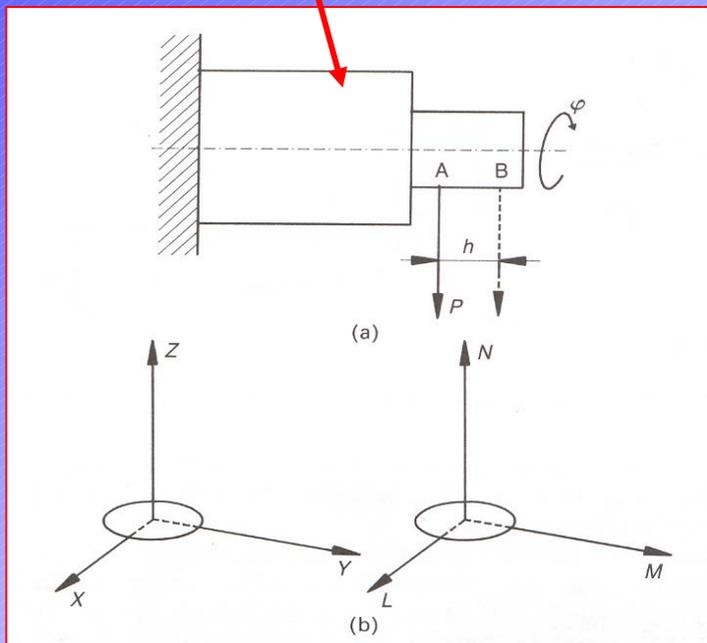
$$L = M = N = 0$$

b) **One-wedge method:** transverse components (X, Y) and bending moments L, M are proportional to Z;

$$N = 0.$$

d) Transverse-load method:
 XM and YL are applied at the same time;
 $Z = N = 0$

Applied by Prof. Dambacher in the '70s to evaluate 130 load cells

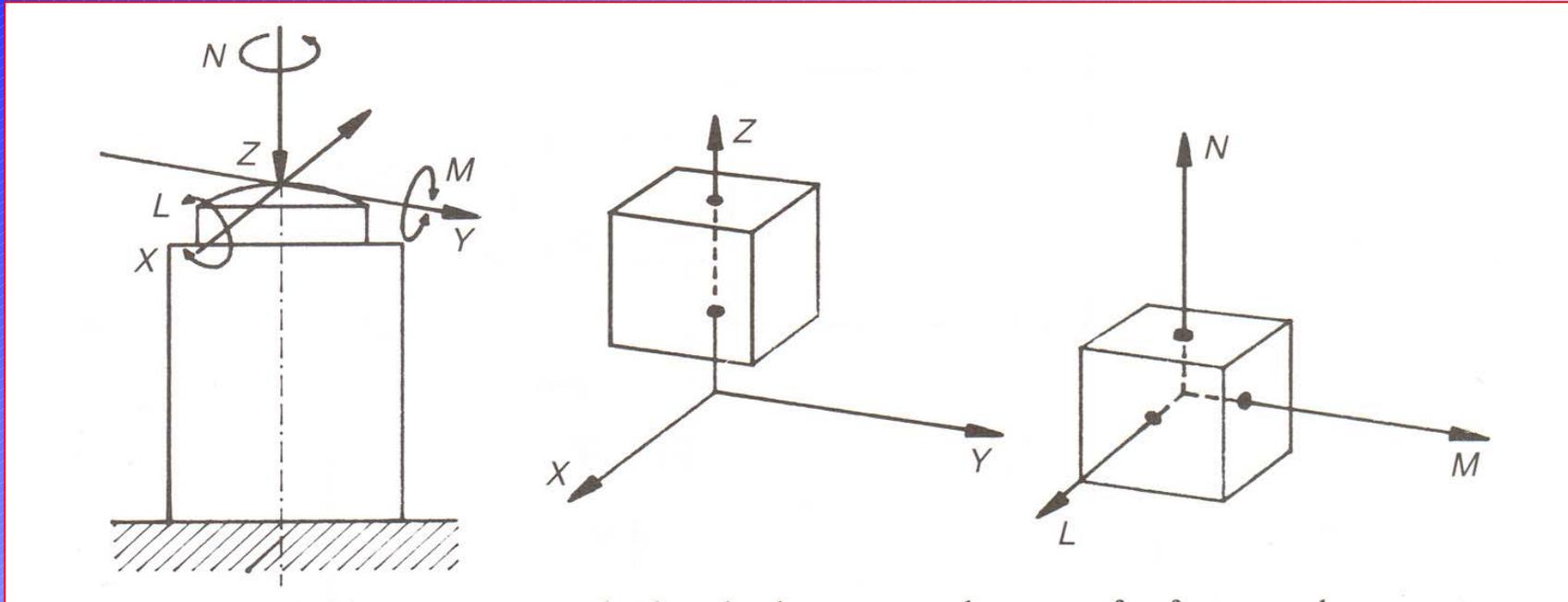


e) Multi-position method:
 bending moments L, M, are proportional to Z;
 $X = Y = N = 0$

Applied by Prof. Levi

Levi R 1967 Drill press dynamometers. Int. J. Mach. Tool Des. Res. 7, 269-87

c) **multi-component method:** independent or contemporary applications of the six components of the force tensor is possible.



Takada R, Ono K, Ogata K, Kusaki T 1988 An analysis of errors on 6-component force/moment calibration machines. *Proc. 11th IMEKO World Conf. Sensors sector*. IMEKO, Budapest, pp. 121-30

Yoshida T 1984, Six-component force transducer and its applications. *Proc. 10th IMEKO Conf. Measurement of Force and Mass*. IMEKO, Budapest

The output signal of a dynamometer as a function of applied components can be adequately approximated in fact by a **second-order polynomial of the type:**

$$y_n = A_o + \sum_{i=1}^6 A_i x_i + \sum_{i=1}^6 \sum_{j=1}^6 A_{ij} x_i x_j$$

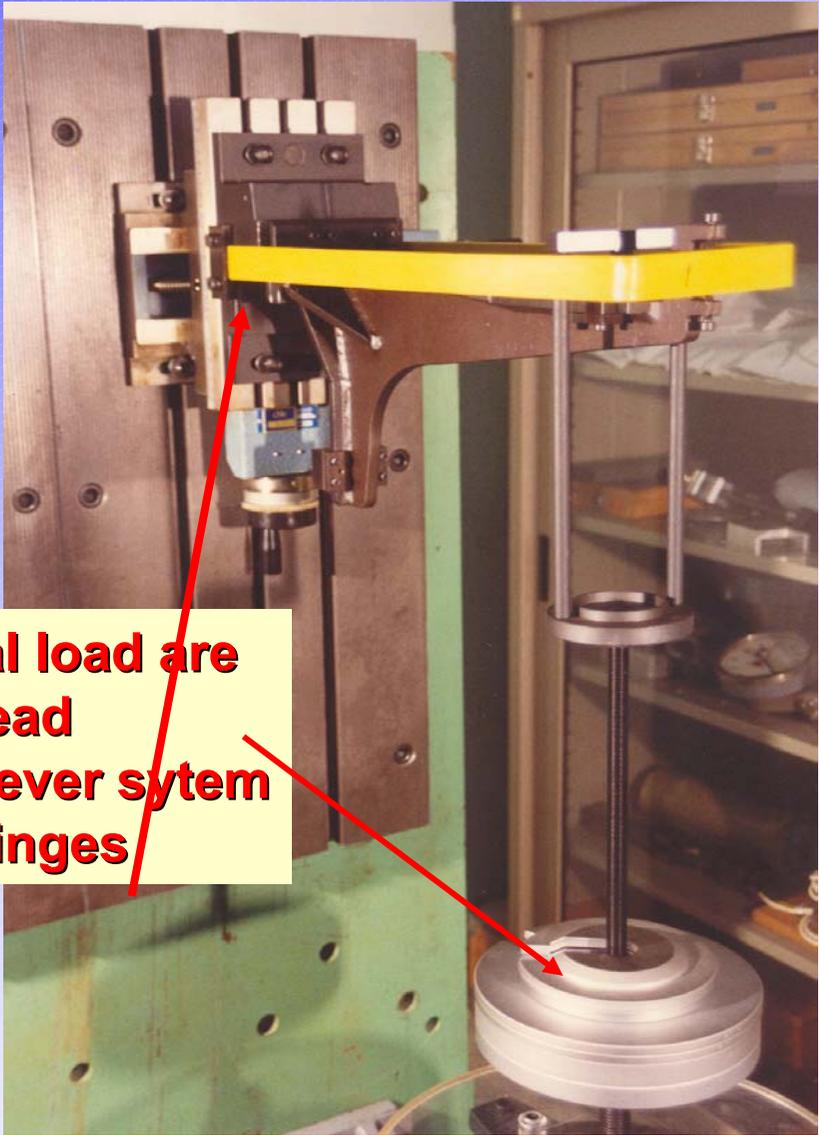
Only with a 6-component ~calibration system can all the twenty seven terms in equation be determined

INRIM CALIBRATION SYSTEMS





The horizontal load are realised by dead weights and lever sytem with elastic hinges



LEVER SYSTEM WITH ELASTIC HINGES

CALIBRATION

- K ratio=2
- Sensitivity
- Repeatability
- Reproducibility

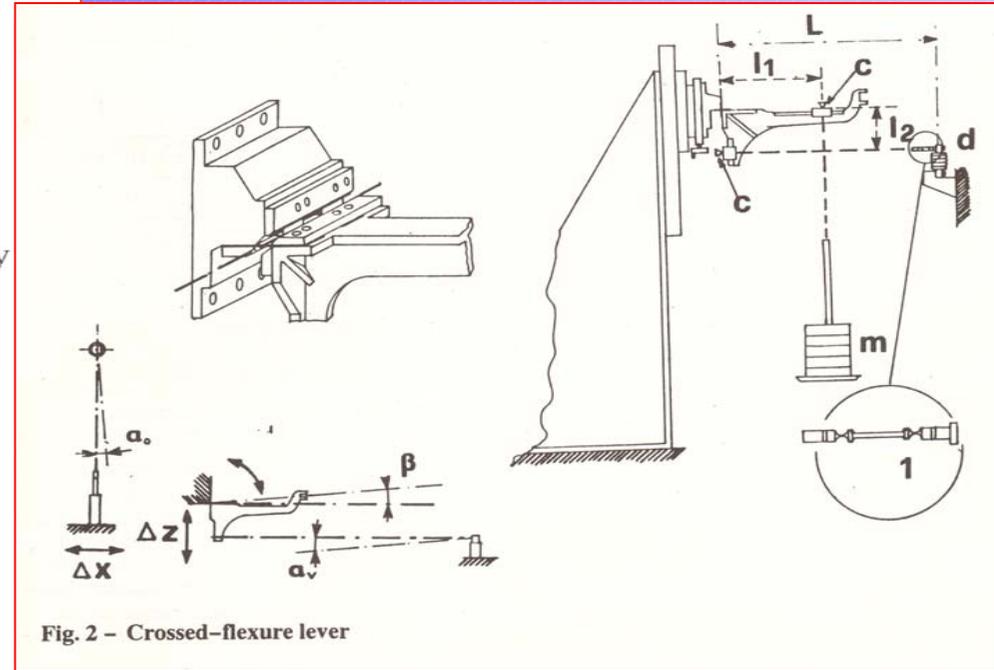
INFLUENCE OF SEVERAL PARAMETERS ON K RATIO

- Tilt angle
- Load level
- Errors in vertical and horizontal positions

DETERMINATION OF APPARENT WEIGHT

- Direct methods
- Indirect meth.

BALANCE OF APPARENT WEIGHT



	LEVER	AIR-BEARING	PULLEY
THRESHOLD	0,5 g/digit	1 g/digit	1 g/digit
REPEATABILITY	$4 \cdot 10^{-5}$	$5 \cdot 10^{-5}$	$6 \cdot 10^{-5}$
VERTICAL DISPLACEMENT EFFECT	$\pm 4 \cdot 10^{-5} / \text{mm}$	$< (\pm 1 \cdot 10^{-5}) / \text{mm}$	$< (\pm 1 \cdot 10^{-5}) / \text{mm}$
HORIZONTAL DISPLACEMENT EFFECT	$(\pm 1,3 \cdot 10^{-5}) / \text{mm}$	$(\pm 2 \cdot 10^{-5}) / \text{mm}$	$(\pm 2,5 \cdot 10^{-5}) / \text{mm}$
ROTATION EFFECT	-	$< 3 \cdot 10^{-4}$	$< 3 \cdot 10^{-3}$
LOADING EFFECT	ratio proportional to applied load	pressure proportional to applied load	friction proportional to applied load
REPRODUCIBILITY	$1 \cdot 10^{-4}$	$3 \cdot 10^{-4}$	$5 \cdot 10^{-4}$
OPERATION CONDITIONS	precise adjustment	needs a pressure source	normal
RELATIVE RATIO	3	2	1
ADVANTAGE	highest accuracy	easy installation high accuracy	lowest cost very easy installation
DISADVANTAGE	highest cost precise adjustment	needs a pressure source	bigger rotation effect threshold changes with load

DIFFERENT EXPERIMENTAL CALIBRATION PLANE and RESULTS EVALUATION

5.2 Loading combination in the sequential calibration method.

Sequential or direct calibration method

DATE	TESTS		Temp. °C	Atm. pressure (mbar)
	COMPONENT	LOADS		
28.2	L	-16; -8; 8; 16 kgf m	20.5 (± 0.2)	988.4
2.3	M	-16; -8; 8; 16 kgf m		987.9
3.3	X	-20; -16; -8; 8; 16; 20 kgf		991.3
23.2	Y	-20; -16; -8; 8; 16; 20 kgf		988.4
1.3	N	-4.8; -3.2; -1.6; 1.6; 3.2; 4.8 kgf m		992.3
1.3	ZN	Z = 4000-8000 kgf		989.1
		N = ±(1.6-3.2-4.8)kgf m		
28.2	ZL	Z = 4000-8000 kgf		988.4
		L = ±(8-16)kgf m		
2.3	ZM	Z = 4000-8000 kgf		992.3
		M = ±(8-16)kgf m		
3.3	ZX	Z = 4000-8000 kgf		988.4
		X = 8-16-20 kgf		
25.2	ZY	Z = 4000-8000 kgf		989.1
		Y = ±(8-16-20)kgf		

122 test condition

A complete calibration of a multicomponent dynamometer (namely, a complete determination of all the a_j and a_{ij} coefficients) by these methods employed at the ONERA and IMGIC laboratories (IMGIC2) requires separate and independent applications of the six main components, i.e., the three orthogonal forces (X, Y, Z), the three moments (L, M, N), and the 15 cross combinations.

Factorial Methods and

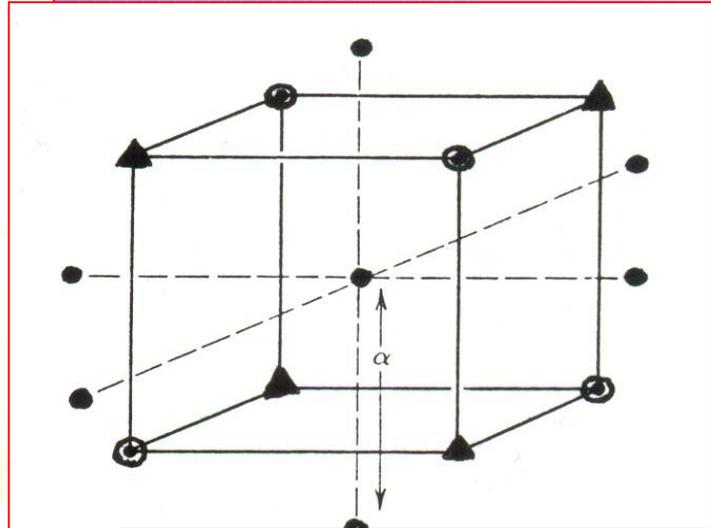


...step-by-step regression analysis

5.4 Combinations for a Central Composite second-order Rotatable design with 6 variables.

x_1	x_2	x_3	x_4	x_5	x_6	x_1	x_2	x_3	x_4	x_5	x_6
-1	-1	-1	-1	-1	-1	-1	1	-1	1	1	1
1	-1	-1	-1	-1	1	1	1	-1	1	1	-1
-1	1	-1	-1	-1	1	-1	-1	1	1	1	1
1	1	-1	-1	-1	-1	1	-1	1	1	1	-1
-1	-1	1	-1	-1	1	-1	1	1	1	1	-1
1	-1	1	-1	-1	-1	1	1	1	1	1	-1
-1	1	1	-1	-1	-1	-2.378	0	0	0	0	0
1	1	1	-1	-1	1	2.378	0	0	0	0	0
-1	-1	-1	1	-1	1	0	-2.378	0	0	0	0
1	-1	-1	1	-1	-1	0	2.378	0	0	0	0
-1	1	-1	1	-1	-1	0	0	-2.378	0	0	0
1	1	-1	1	-1	1	0	0	2.378	0	0	0
-1	-1	1	1	-1	-1	0	0	0	-2.378	0	0
1	-1	1	1	-1	1	0	0	0	2.378	0	0
-1	1	1	1	-1	1	0	0	0	0	-2.378	0
1	1	1	1	-1	-1	0	0	0	0	2.378	0
-1	-1	-1	-1	1	1	0	0	0	0	0	-2.378
1	-1	-1	-1	1	-1	0	0	0	0	0	2.378
-1	1	-1	-1	1	-1	0	0	0	0	0	0
1	1	-1	-1	1	1	0	0	0	0	0	0
-1	-1	1	-1	1	-1	0	0	0	0	0	0
1	-1	1	-1	1	1	0	0	0	0	0	0
-1	1	1	-1	1	-1	0	0	0	0	0	0
1	-1	1	-1	1	1	0	0	0	0	0	0
-1	-1	-1	-1	-1	-1	0	0	0	0	0	0

The component values assumed in the STARS was optimized according to the work of Box-Hunter



The spatial distribution of the tests concerning a three component CCRD is schematized in the Figure

Table gives the combinations of one of the designs adopted at IMGC

Table 5.3 Loading combinations with a complete factorial design of the 2^5 type repeated for 3 levels of axial load Z.

Reference	N°	$x_2 = L$	$x_3 = M$	$x_4 = N$	$x_5 = X$	$x_6 = Y$	$x_1 = Z$
-----------	----	-----------	-----------	-----------	-----------	-----------	-----------

I	05	0	0	0	0	0	-1,0,1
(1)	06	-1	-1	-1	-1	-1	
ab	07	1	1	-1	-1	-1	
ace	08	1	-1	1	-1	1	
bce	09	-1	1	1	-1	1	
	10	0	0	0	0	0	
ade	11	1	-1	-1	1	1	
bde	12	-1	1	-1	1	1	
cd	13	-1	-1	1	1	-1	
abcd	14	1	1	1	1	-1	
	15	0	0	0	0	0	

II	20	0	0	0	0	0	-1,0,1
a	21	1	-1	-1	-1	-1	
b	22	-1	1	-1	-1	-1	
ce	23	-1	-1	1	-1	1	
abce	24	1	1	1	-1	1	
	25	0	0	0	0	0	
de	26	-1	-1	-1	1	1	
abde	27	1	1	-1	1	1	
acd	28	1	-1	1	1	-1	
bcd	29	-1	1	1	1	-1	
	30	0	0	0	0	0	

III	35	0	0	0	0	0	-1,0,1
ae	36	1	-1	-1	-1	1	
be	37	-1	1	-1	-1	1	
c	38	-1	-1	1	-1	-1	
abc	39	1	1	1	-1	-1	
	40	0	0	0	0	0	
d	41	-1	-1	-1	1	-1	
abd	42	1	1	-1	1	-1	
acde	43	1	-1	1	1	1	
bcde	44	-1	1	1	1	1	
	45	0	0	0	0	0	

Reference	N°	$x_2 = L$	$x_3 = M$	$x_4 = N$	$x_5 = X$	$x_6 = Y$	$x_1 = Z$
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IV	50	0	0	0	0	0	-1,0,1
e	51	-1	-1	-1	-1	1	
abe	52	1	1	-1	-1	1	
ac	53	1	-1	1	-1	-1	
bc	54	-1	1	1	-1	-1	
	55	0	0	0	0	0	
ad	56	1	-1	-1	1	-1	
bd	57	-1	1	-1	1	-1	
cde	58	-1	-1	1	1	1	
abcde	59	1	1	1	1	1	
	60	0	0	0	0	0	

DOUBLE STAR	70	0	0	0	0	0	-1,0,1
	71	-4	0	0	0	0	
	72	-1	0	0	0	0	
	73	1	0	0	0	0	
	74	4	0	0	0	0	
	75	0	0	0	0	0	
	76	0	-4	0	0	0	
	77	0	-1	0	0	0	
	78	0	1	0	0	0	
	79	0	4	0	0	0	
	80	0	0	0	0	0	
	81	0	0	-3	0	0	
	82	0	0	-1	0	0	
	83	0	0	1	0	0	
	84	0	0	3	0	0	
	85	0	0	0	0	0	
	86	0	0	0	-2.5	0	
	87	0	0	0	-1	0	
	88	0	0	0	1	0	
	89	0	0	0	2.5	0	
90	0	0	0	0	0		
91	0	0	0	0	-2.5		
92	0	0	0	0	-1		
93	0	0	0	0	1		
94	0	0	0	0	2.5		

By using the STEP BY STEP REGRESSION ANALYSIS (from Efroymsen)...

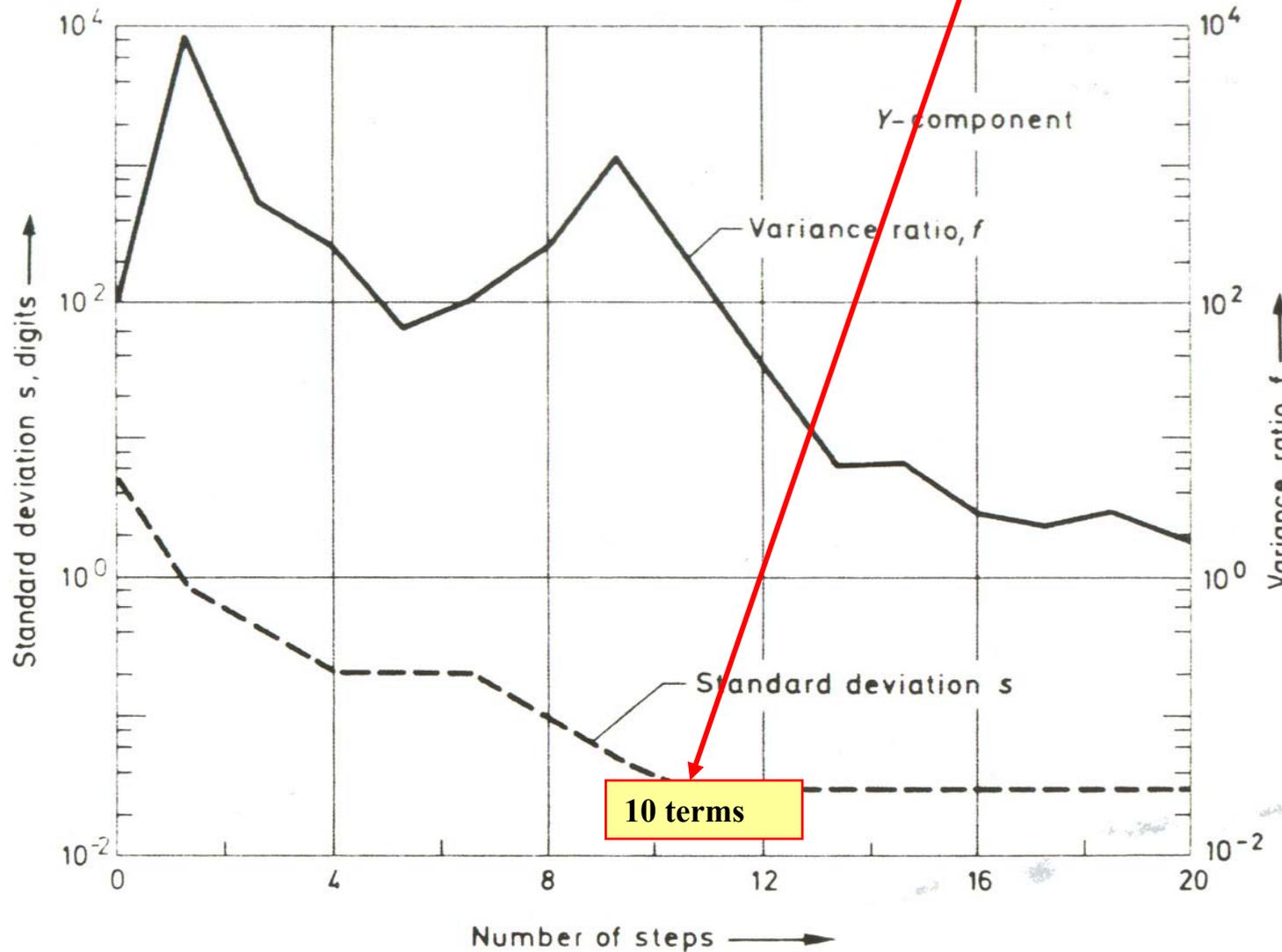
		COEFFICIENT VALUE AT STEP NUMBER								
Variable	Coeff.	1	2	3	4	5	6	7	8	9
O_2	a_6	8.25×10^{-4}	8.13×10^{-4}	6.27×10^{-4}	6.99×10^{-4}	6.99×10^{-4}	7.00×10^{-4}	7.02×10^{-4}	7.02×10^{-4}	7.00×10^{-4}
O_3	a_7		-5.66×10^{-4}	-5.46×10^{-4}	-7.63×10^{-4}	-7.63×10^{-4}	-7.65×10^{-4}	-7.65×10^{-4}	-7.65×10^{-4}	-7.65×10^{-4}
O_1	a_5			-5.10×10^{-4}	-6.97×10^{-4}	-7.01×10^{-4}	-7.01×10^{-4}	-7.01×10^{-4}	-7.01×10^{-4}	-7.02×10^{-4}
\bar{v}	a_1				-9.00×10^{-4}	-8.98×10^{-4}	-8.97×10^{-4}	-8.97×10^{-4}	-8.96×10^{-4}	-8.94×10^{-4}
$\bar{v}O_1$	a_{1-5}					4.47×10^{-8}	4.52×10^{-8}	3.43×10^{-8}	3.23×10^{-8}	3.25×10^{-8}
$\bar{v}O_3$	a_{1-7}						3.07×10^{-8}	3.12×10^{-8}	2.89×10^{-8}	2.89×10^{-8}
$\bar{v}O_2$	a_{1-6}							-3.18×10^{-8}	-3.10×10^{-8}	-3.08×10^{-8}
\bar{v}^2	a_{1-1}								-1.70×10^{-8}	-1.69×10^{-8}
v_2	a_3									-3.35×10^{-4}
Constant	a_0	9.2×10^{-4}	-2.5×10^{-2}	-4.1×10^{-2}	-3.8×10^{-2}	-2.3×10^{-2}	-1.3×10^{-2}	9.8×10^{-3}	5.7×10^{-4}	5.7×10^{-4}

.....the evolution of the coefficients of the regression equation is determined at each step.

The relevant ratios of the regression equation coefficients to the standard deviations at the corresponding steps are determined as well

Variable	Coeff.	RATIO VALUE AT STEP NUMBER								
		1	2	3	4	5	6	7	8	9
O_2	a_6	1.28×10^1	1.56×10^1	1.46×10^1	1.77×10^2	2.49×10^2	3.35×10^2	7.83×10^2	9.59×10^2	1.00×10^3
O_3	a_7		-1.07×10^1	-1.34×10^1	-1.92×10^2	-2.70×10^2	-3.62×10^2	-8.49×10^2	-1.04×10^3	-1.15×10^3
O_1	a_5			-1.19×10^1	-1.71×10^2	-2.40×10^2	-3.23×10^2	-7.54×10^2	-9.24×10^2	-9.76×10^2
\bar{v}	a_1				-1.56×10^2	-2.19×10^2	-2.93×10^2	-6.87×10^2	-8.37×10^2	-8.88×10^2
$\bar{v}O_1$	a_{1-5}					1.41×10^1	1.91×10^1	3.20×10^1	3.60×10^1	4.02×10^1
$\bar{v}O_3$	a_{1-7}						1.27×10^1	$+3.03 \times 10^1$	3.32×10^1	3.67×10^1
$\bar{v}O_2$	a_{1-6}							-3.01×10^1	-3.58×10^1	-3.93×10^1
\bar{v}^2	a_{1-1}								1.01×10^1	-1.11×10^1
v_2	a_3									-6.81×10^1

Cut criteria are better defined in the diagram of fig, which gives the behavior of experimental values of **variance ratio F** and **standard deviation S_y** for component **Y**.

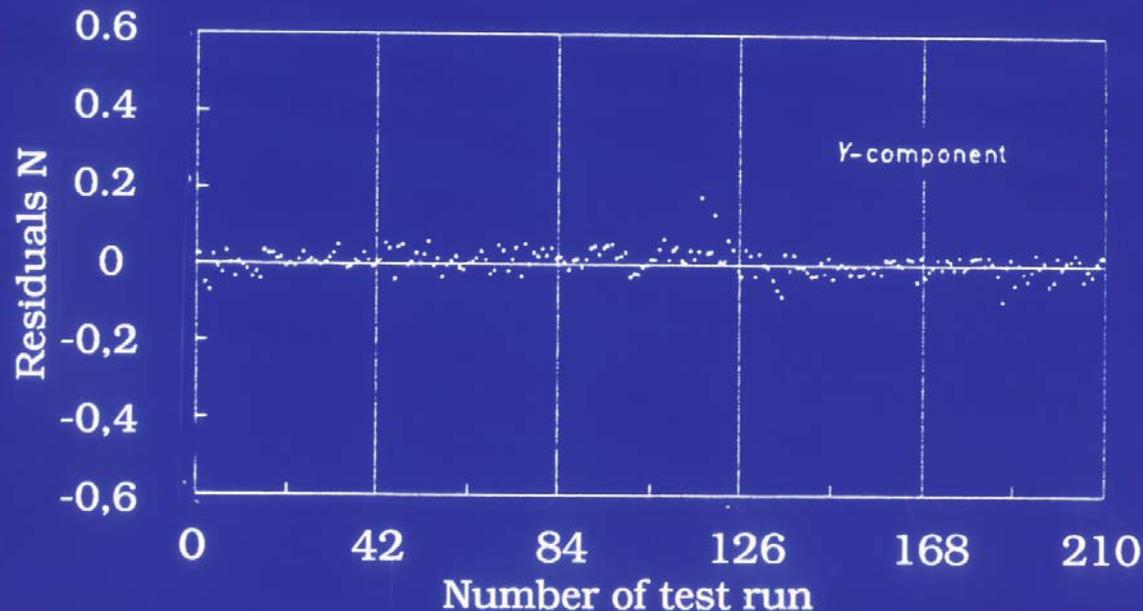


For purpose of economy; only the terms giving meaningful contributions are maintained in the equations.

Forces	Terms
$X = a_0 + a_1 \bar{V} + a_2 V_1 + a_4 V_3 + a_5 H_1 + a_6 H_2 + a_7 H_3$ $+ a_{1,1} \bar{V}^2 +$ $+ a_{1,3} \bar{V}V_2 + a_{1,4} \bar{V}V_3 + a_{1,5} \bar{V}H_1 + a_{1,6} \bar{V}H_2 + a_{1,7} \bar{V}H_3$	linear square crossed
$Y = a_0 + a_1 \bar{V} + a_2 V_1 + a_3 V_2 + a_5 H_1 + a_6 H_2 + a_7 H_3$ $+ a_{1,1} \bar{V}^2$ $+ a_{1,3} \bar{V}V_2 + a_{1,5} \bar{V}H_1 + a_{1,6} \bar{V}H_2$	linear square crossed
10 terms	
Moments	
$L = a_0 + a_1 \bar{V} + a_2 V_1 + a_4 V_3 + a_5 H_1 + a_6 H_2 + a_7 H_3$ $+ a_{1,1} \bar{V}^2$ $a_{1,2} \bar{V}V_1 + a_{1,3} \bar{V}V_2 + a_{1,5} \bar{V}H_1 + a_{1,6} \bar{V}H_2$	linear square crossed
$M = a_0 + a_1 \bar{V} + a_2 V_1 + a_4 V_3 + a_5 H_1 + a_6 H_2 + a_7 H_3$ $a_{1,1} \bar{V}^2$ $+ a_{1,3} \bar{V}V_2 + a_{1,4} \bar{V}V_3 + a_{1,5} \bar{V}H_1 + a_{1,6} \bar{V}H_2 + a_{1,7} \bar{V}H_3$	linear square crossed
$N = a_0 + a_1 \bar{V} + a_3 V_2 + a_5 H_1 + a_6 H_2 + a_7 H_3$ $+ a_{1,1} \bar{V}^2$ $+ a_{1,5} \bar{V}H_1 + a_{1,6} \bar{V}H_2 + a_{1,7} \bar{V}H_3$	linear square crossed

With the IMGDC dynamometer only from 9 to 12 coefficients are retained in the equations.

Finally, residuals (differences between calculated and actual values) for 210 testing conditions obtained with different load combinations are quoted in figure which illustrates the behavior of component Y ($Y = 200 \text{ N}$)



Sensitivity Coefficients Matrix

COMPONENTS			Channel V1	Channel V2	Channel V3	Channel H1	Channel H2	Channel H3
LINEAR TERMS	Force	Z	-32.5526	-32.9338	-32.5518	0.5095	0.1634	-0.903
		X	321.1	-158.5	-162.7	-432	171.6	881.4
		Y	6.02	-282.75	276.5	-344.3	-61	411.4
	Moments	L	1	-441.25	441.87	6.25	2	10.03
		M	513.75	-260.6	-254.35	0.4	10.09	-0.6
		N	0	0	0	1098	1093.7	1088
SECOND ORDER TERMS	Square terms 10E-06	Z²	3.44	-8.2	6.08	8.05	-9.96	-
		X²	0	0	0	0	0	-
		Y²	0	0	0	0	0	-
		L²	0	0	0	0	0	-
		M²	0	0	0	0	0	-
		N²	0	0	0	0	0	-
	Rectang. terms 10E-03	ZX	2.17	1.25	1.1	1.975	3.62	-
		ZY	0	-0.635	-0.5	-0.715	-0.45	-
		ZL	0.1	-1.2	-1.562	-0.156	0.1	-
		ZM	2.02	0.94	1.09	2.69	1.25	-
		ZN	0	0	0	2.75	3.125	3

It is interesting to show that the SUM of the VERTICAL CHANNEL (Vi) is not interacting with the other components

Many deadweight Force Standard Machines of the main National Institute of Metrology around the world have been evaluated with the **IMRiM six-component Dynamometers**

Institutes	Year	
N.P.L. (U.K.)	1985	1996
T.N.O. (the Nederland)	1985	
L.G.A.I. (Spain)	1994	2003
N.R.L.M. (Japan)	1994	
N.I.M. (China)	1986	1998
L.N.E. (France)	1985	1995
K.R.I.S.S. (Korea)	1998	
P.T.B. (Germany)	1988	2004
R.P.O. (Finland)	1998	

GENERAL EVALUATION

Intercomparison tests of DWM exhibited sometimes lack of agreement inconsistent with the accuracy of the machines and the estimated uncertainty of measurement.

Such errors, are originated mainly

- **by the presence of unwanted, transversal force (and moment) components, applied by the deadweight machine**
- **the non-zero sensitivity to these components exhibited by the load cell.**

Quantitative evaluation of parasitic load components applied by the machine, and of their single and combined effects on load cell output, are to be obtained in order to reduce uncertainty in force measurement.

However,

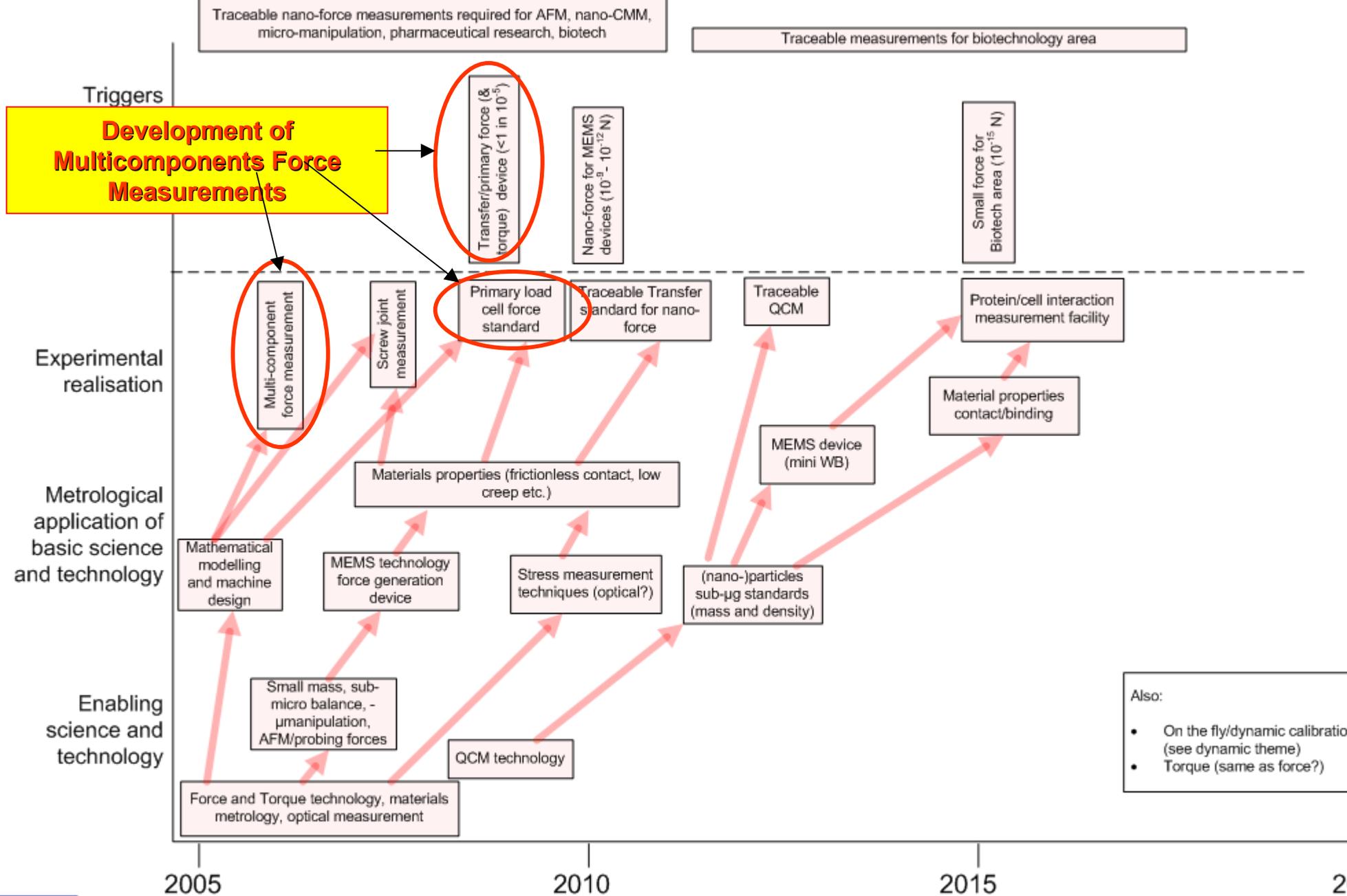
- 1) Transversal load components applied by deadweight machines must be measured, and effective steps taken to limit their magnitude,**
- 2) Measurement of load cell sensitivity to transversal loads is also to be performed systematically.**

In other words it will be necessary in future activities of the Force Working Group

- a) to prescribe methods and means which ensure **uncertainty estimation of parasitic components** in force measurements;
- b) to **define** an accepted **theoretical volume of the parasitic components for the primary standards**;
- c) to **determine (and limit) the sensitivities** to transversal loads and bending moments **for the transfer standards to be used in force comparisons.**

**What was disseminate in the last 30 years by
INRiM?**

**What the future in the multicomponent force
activities: research and intercomparison?**



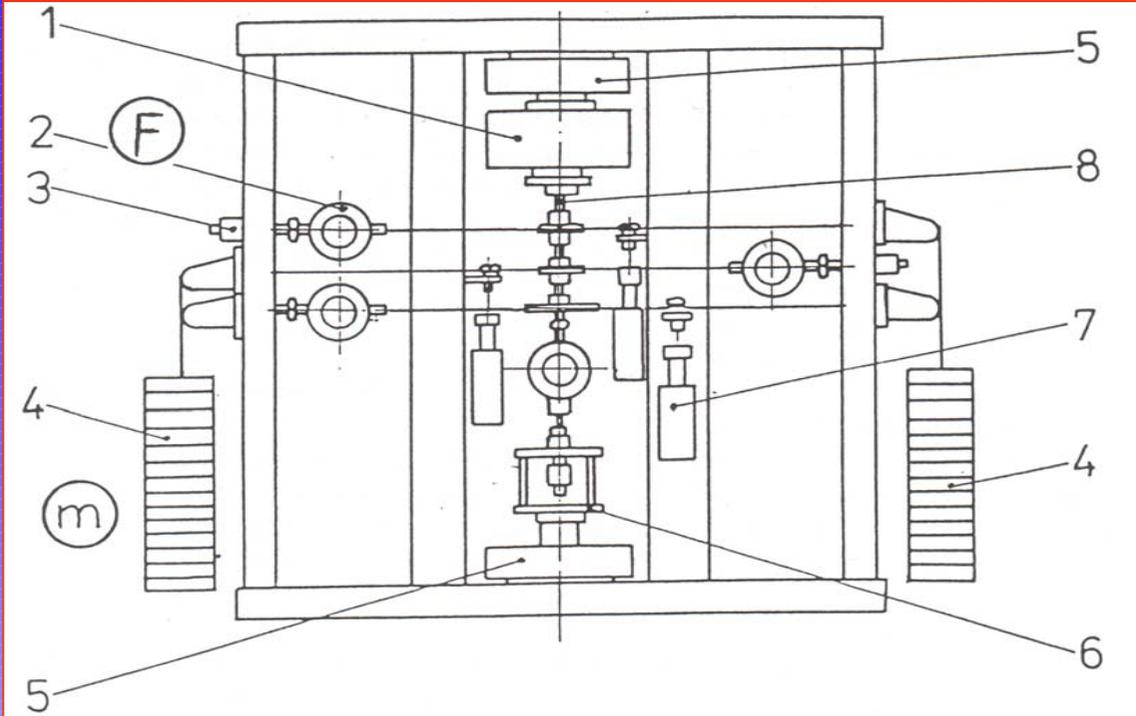


Fig. 1 Force and mass versions for calibration stand

An interesting method was developed by Dae-Im Kang and his collaborators by using build-up system for the evaluation of FSM

Cooperation Ferrero-Dae Im Kang (1996)

NIM CALIBRATION SYSTEM - CHINA

System developed in
China to be used in the
aero-space field (**very
high force and
moment components**)

**Cooperation Bray-Ferrero-Li
QingZhong-Marinari
(dal 1983 al 1995)**

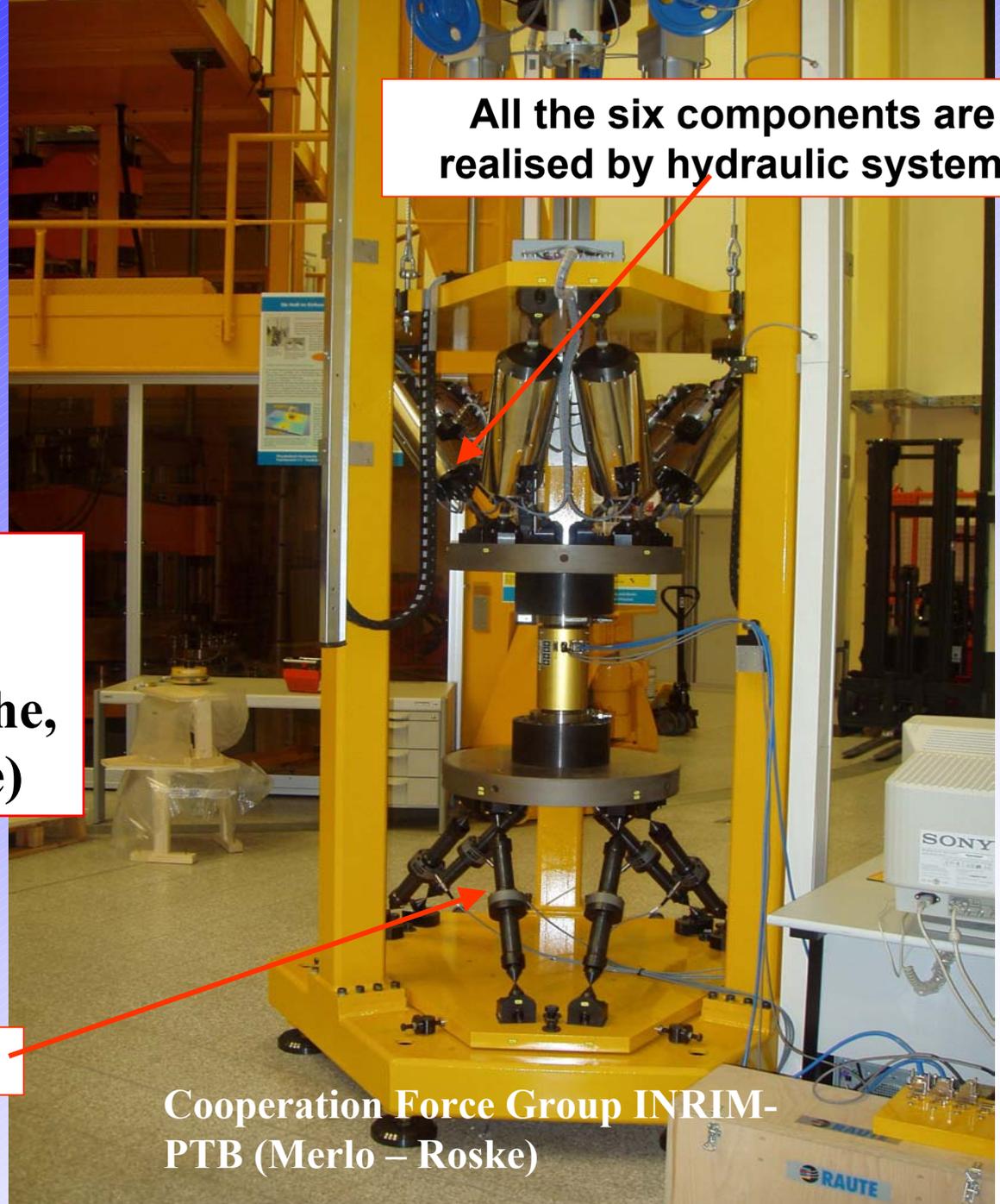


PTB SYSTEM FOR HIGH FORCES

Controllo delle componenti nel tempo in strutture in cemento armato (ponti, dighe, costruzioni in zone sismiche)

...and measured by load cells

All the six components are realised by hydraulic systems



Cooperation Force Group INRIM-PTB (Merlo – Roske)

**Many Thanks ...
for your attention**

